DG Flugzeugbau GmbH 76646 Bruchsal		Technical note No. 800/41	page 1 from 3
Subject	:	Coolant pump, spark plug connectors, mounting of the ga primer valve, retaining spring for pedal adjustment cable revision	
Effectivity	• •	DG-800 DG-800B with engine Solo 2625, DG-808C	
Accomplishment	:	 In case the coolant pump is defective. In case new spark plug connectors are needed. Check prior to next flight. Exchange if necessary unalatest until March. 31. 2013. Only DG-800B: In case the primer valve is defective primer valve is no longer available. Before the next annual inspection, at latest by March When exchanging the coolant hoses. With the 25 hour inspection. Every 50 engine hours. Optional. Check prior to next take-off. Only DG-800B: Before the next annual inspection, at latest primer take primer take primer take-off. Only DG-800B: Before the next annual inspection, at latest primer take-off. Only DG-800B: With the next exchange of the fuel primer take primer take-off. Only DG-808C: Before the next annual inspection, at latest primer take-off. Before the next annual inspection or if one of measu executed (with exemption of the check of instruction 2013. 	e, optional, or if the existing h. 31. 2013 (only DG-808C). at latest by March. 31. 2013. hoses. at latest by March. 31. 2013. res 1 up to 6 has to be
Reason	:	 The existing coolant pump is no longer available. A tested. A different coolant mix. must be used with th Only DG-808C: The failure message "water pump" DEI-NT. Instead of the spark plug connectors type "Bosch" c be used. At the mounting point of the gas strut at the powerpl shall prevent the gas strut from disconnecting from t of the gas strut fails. Only DG-800B: The primer valve which has been d may be used for the DG-800B. Only DG-808C: During the revision of the electrica was detected that a fuse for the fire warning light is 16. It is acceptable to exchange the coolant only every 6 coolant hoses. It is not necessary to exchange the spark plugs and th only every 25 hours with the 25 hour inspection. Operational experience has shown that the upper drivreplaced every 50 engine hours (this limitation was o MIM). A modified mounting of the propeller-brake retainin wear of the levers and the connecting rod. If the spring which pulls the pedal adjustment cable v that it may hook into the trim release lever at the correar position. Only DG-800B: The check if the fuel cock is open i placard "Pre-flight inspection". Only DG-800B: Replace the plastic fuel line fittings next exchange of the fuel hoses. Only DG-800B: Replace the plastic fuel line fittings next exchange of the fuel hoses. 	the new pump. must be deactivated in the connectors type "PVL" may lant a large washer S12/1 he powerplant if the rod end esigned for the DG-808C I system of the DG-808C it not installed. years when replacing the he fuel filter every year but we belt rollers must not be ponly given in theDG-808C g rubber cord may reduce the tight is defective or not won't be pulled to the front so ntrol stick with pedals in a as missing on the cockpit- s with metal fittings at the ed in some DG-808C will be r looses its transparency the tion which leads to a wrong

DG Flugzeugbau GmbH 76646 Bruchsal	Technical note No. 800/41	page 2 from 3
	 14. Revision of flight-, maintenance- and repair main The following DG Service Infos have been incomanuals: a) SI 48-02 Check of the coolant pump. b) SI 64-07 Carburettor type Mikuni B38 new v carburettor membrane which has already been c) SI 69-10 Adjustment of propeller after jump of MM). 	rporated into the maintenance ersion: Reinstallation of a n used.
Instructions :	 Coolant pump: Modification and exchange of co 74-12. Only DG-808C: New control unit software Vers Use of PVL spark plug connectors (part No. see Check if a large washer S12/1 is installed, see d diagram 10 DG-808C. The outside diameter sho diameter is smaller the washer must be replaced Only DG-800B: Exchange of primer valve and according to working instruction No. 1 for TN8 Installation of a fuse for the fire warning light at No. 2 for TN800/41. Exchange of coolant (see MM section 4.16) tog hoses. Exchange of spark plugs and fuel filter (see MM engine time. Exchange drive belt rollers only when worn. Modify installation of the propeller-brake retain diagram 13b. Check if the handle of the pedal adjustment cab that it can't hook into the trim release lever at the 	sion 1.6 must be installed. MM section 8.1). iagrams 10a and 10 DG-800B and ould be 22 mm (0.87 in.). If the l. modification of mounting 00/41. ccording to working instruction ether with exchange of coolant 4 section 3.5.1) only every 25 h ing rubber cord according to MM le will be pulled to the front so according tick, even with pedals
	 in a rear position. If this is not the case exchang according to MM section 1.3.6. This check is not inspection program in the flight manual. 11. Only DG-800B: Exchange the cockpit-placard suitable placard "Fuel cock open?" below the exitable placard "Fuel cock open?" belo	ow incorporated in the daily "Preflight inspection" or install a kisting placard. trings with metal fittings according efer to diagram 11b or 11d in nk sensor according to working
	 14. Manual revision: DG-800B: Exchange the complete Flight and many changes and a new format of the MM page existing manuals. Respect the changes marked in the right hand mAFM: 0.1, 1.2, 1.5, 1.6, 2.2, 2.5, 2.6, 2.9 ÷ 2.12 4.22, 4.25 ÷ 4.28, 5.3, 5.8 ÷ 5.10, 6.3, 7.2, 7.14, MM: 0.13, 0.14, 1.2, 1.4, 1.5, 1.7 ÷ 1.11, 1.13, 2.1, 2.4, 3.1 ÷ 3.6, 3.8 ÷ 3.10, 4.1 ÷ 4.3, 4.7 ÷ 4.1 4.25 ÷ 4.31, 5.1, 5.2, 6.1 ÷ 6.3, 8.1 ÷ 8.5, 9.2, di 11, 11a, 11b, 11d, 12a, 13b, 14 (14a removed), 8M210, W59, SI 69-10 	e No.'s are different to the argin on the following pages: , 3.4, 3.7, 4.5, 4.8, 4.9, 4.11, 4.15, 7.16 ÷ 7.18, 8.7, 9.1 ÷ 9.5 1.14 ÷ 1.16, 1.18, 1.26, 1.27, 1.30, 11, 4.13, 4.15, 4.16, 4.19 ÷ 4.23, agrams 1 ÷ 6, 7, 7a, 8, 9, 10a,10b,
	DG-808C: Exchange the following manual pag 2012 marked with TN800/41. Respect the chang margin AFM: 0.1, 0.3 ÷ 0.5, 1.2, 1.5, 2.6, 2.8, 2.9, 3.4, 4.20, 4.24, 4.26, 4.27, 4.29, 5.14, 6.2, 6.4, 6.7, 6 7.15, 7.18, 7.20, 7.22, 7.23	ges marked in the right hand 4.4 ÷ 4.6, 4.8, 4.9, 4.11, 4.15,

DG Flugzeugbau GmbH 76646 Bruchsal	Technical note No. 800/41	page 3 from 3
	MM: 0.1, 0.3 ÷ 0.12, 1.3, 1.5, 1.6, 1.8, 1.10, 1.11, 1.14 ÷ 1. 1.30, 2.1, 2.2, 2.5, 2.6, 2.7, 3.1, 3.3 ÷ 3.11, 4.1 ÷ 4.3, 4.6 ÷ 4.18 ÷ 4.23, 4.25 ÷ 4.31, 5.1, 5.2, 6.1 ÷ 6.3, 7.2, 8.1 ÷ 8.4, 9 8, 9, 10, 11, 11d, 12, 13, 13b, 14, 17, 14a removed, 8M110, 8E250, SI 69-10 We recommend to print out the complete manuals instead of	4.11, 4.13, 4.15, 4.16, 9.2, diagrams 1 ÷ 5, 5a,
	due to the many changes.	r exchanging pages
	DG-800B and DG.808C: The DG-800B repair manual is v Exchange the complete manual as due to the many changes page No.'s are different to the existing manual. Respect the changes marked in the right hand margin on the $1.1, 2.1, 4.1 - 4.3, 5.4, 6.1 \div 6.3$	and a new format
Material	 Service Info 74-12 with the material listed therein 45002085 PVL spark plug connectors 5kΩ (DG part. No. 4 40050121 Large washer S12/1 Working instruction No. 1 for TN800/41 with the material I Working instruction No. 2 for TN800/41 with the material I Working instruction No. 2 for TN800/41 with the material I / / 53200622 1 item clevis pin 6H11x22x19,5 DIN 1434 St zt 60510951 2 items Ty-raps 92x1,6 30091130 70 cm rubber cord dia. 2mm if defective: 60000295 tension spring 55x12x1 60000226 TSV8 MS, replacement for TS8 and YS8 60000225T-Red. 8-6-8, replacement for TRS 8-6-8 60000221 8/6 Stahl, replacement for T GRS 8-6 (only up to 13. Working instruction No. 3 for TN800/41 with the material I 	isted therein isted therein n ser.no. 8-102)
	 13. Working instruction No. 5 for TN800/41 with the material 1 14. DG-800B: Flight manual DG-800B Revision 19 May 2012, Maintenance manual DG-800B Revision 25 May 2012, Repair manual DG-800B Revision 4 May 2012 DG-808C: Manual pages see instruction 14 and Repair mar Revision 4 May 2012 or Flight manual DG-808C Revision 2 May 2012, Maintenance manual DG-808C Revision 4 May 2012, Repair manual DG-800B 4 May 2012 	
-	influence negligible	
Remarks :	Instructions No. 1-3 and 5-14 may be executed by the pilot/own From instruction 13 only the determination of the type of the set by the pilot/owner, not the exchange of the sensor. The instructions are to be inspected and entered in the aircraft lo inspector at latest with the next annual inspection.	nsor may be performed
	 Instructions No. 4 and the exchange of the sensor from instruction EASA countries: The actions have to be performed in a Parrorganisation, or in a Part M, Subpart F approved organisation regulations of the Part M and released according to M.A.80 Non EASA countries: The actions have to be performed in All instructions are to be inspected and entered in the aircrainspector. 	t -145 approved on according to the 1. a licensed workshop.
Bruchsal, date: 24. July 2012		
Author: W. Dirks	Modifications approved by EASA Date 14.09.2012 under Approval No. 10041426	
Wilhelm Da		