No.	Page	Description	Date
12	$0.2 \div 0.9, 0.9a, 0.10 \div 0.12,$	Manual revision TN1000/18	February
	1.2, 1.5, 1.11, 1.14, 1.15,		2011
	$1.27 \div 1.29, 1.33, 2.1,$		
	$2.4 \div 2.6, 3.3, 4.8, 5.1, 6.1,$		
	6.2, 6.4, 8.3, 9.2, diagr. 1,		
	9, 11, encl. 4 pages 1, 2, 2a,		
	3, Z193, SI 67-07, remove		
	5EP50		
13	0.6, diagrams 8 and 9	Wheel brake TN1000/21	July 2011
14	$0.2 \div 0.7, 0.10 \div 0.12, 1.3,$	Manual revision TN 1000/24,	October
	1.5, 1.11, 1.16, 1.29, 1.30,	New type 12V sockets and	2014
	2.1, 2.6, 3.1, 4.6, 4.10,	plugs,	
	4.12, 4.13, 4.19, 4.20, 6.1,	Changes due to TN 4603-14 of	
	7.1, 8.2, 8.3,	the Solo company (Exchange	
	diagrams: 2, 3, 9,	of the axle of the upper drive	
	Enclosure 4 pages: 4, 7	belt pulley on pages 4.19 and	
		4.20)	
15	0.2, 0.3 - 0.6, 0.9a, 0.11,	Propeller adapter ring with	August
	1.18, 3.5, 3.8, 4.20, 4.20a,	elastomeric damper	2015
	4.26, 4.28, 8.1 – 8.3	TN 1000/26	
	diagram 13, diagram 15,	Alternative mechanical fuel	
	diag0ram 15a	pump TN 1000/28	
16	0.2 -0.6, 0.11, 0.12, 2.1,	Manual revision	July 2017
	2.2, 3.7, 6.3, 8.2, 8.3,	TN1000/32	
	diagram 2		

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0.2 List	of effectiv	ve pages			
Section	page	issued	replaced/	replaced/	replaced/
0	0.0	June 2005	January 2007		
	0.1	see manual	amendments		
	0.2		**		
	0.3		**		
	0.4		"		
	0.5		"		
	0.6		"		
	0.7	June 2005	January 2007	Febr. 2011	Oct. 2014
	0.8	"	Febr. 2011		
	0.9	"	Febr. 2011		
	0.9a	Febr. 2011	August 2015		
	0.10	"	Febr. 2008	March 2008	Oct. 2008
			Nov. 2008	Febr. 2011	Oct. 2014
	0.11	"	Febr. 2011	Oct. 2014	August 2015
			July 2017		e
	0.12	"	January 2007	Febr. 2011	Oct. 2014
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	1.11	"	Febr. 2011	Oct. 2014	
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	1.14	**	March 2008	Febr. 2011	
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	1.22	"	October 2006		
	1.23	"	October 2006		
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	1.31	"	May 2008		
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	1.33	"	Febr. 2011		
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			July 2017		
	2.2	"	July 2017		
	2.3	"			
	2.4	"	Febr. 2011		
	2.5	"	Febr. 2011		
	2.6	**	May 2008	Febr. 2011	Oct. 2014
3	3.1	June 2005	Oct. 2014		
	3.2	"			
	3.3	"	Febr. 2011		
	3.4	"			
	3.5	"	August 2015		
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	3.7	"	July 2017		
	3.8	"	August 2015		
	3.9	"	C		
4	4.1	June 2005			
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	4.6	"	Oct. 2014		
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	4.12	"	Oct. 2014		

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	4.13	"	Oct. 2014		
	4.14	"			
	4.15	"			
	4.16	"	January 2007		
	4.17	"	January 2007		
	4.18	"	May 2008		
	4.19	"	Oct. 2014		
	4.20	"	Oct. 2014	August 2015	
	4.20a	August 2015			
	2.21	"	January 2007		
	4.22	"			
	4.23	"			
	4.24	"			
	4.25	"			
	4.26	"	August 2015		
	4.27	"			
	4.28	"	January 2007	August 2015	
	4.29	"			
	4.30	"			
	4.31	"			
5	5.1	June 2005	Febr. 2011		
	5.2	"			
6	6.1	June 2005	Febr. 2011	Oct. 2014	
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	6.3	"	July 2017		
	6.4	"	Febr. 2011		
7	7.1	June 2005	Oct. 2014		
8	8.1	June 2005	January 2007	August 2015	
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			August 2015	July 2017	
	8.4	"	January 2007	-	
9	9.1	June 2005	2		
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19	March 2008				
20	Nov. 2008				
21	Nov. 2008	Oct. 2010			
22	Nov. 2008				
5EP34	25.01.90				
5V18	14.10.94				
10FW2	05.10.99				
10E4 issue A	28.10.08				
10E4 issue E	8.10.10				
10E102	14.09.05	5.12.05	26.01.06		
10E103	24.06.05				
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SI 67-07	5.11.2007				
Z193	4.11.2009				

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0.4 Airworthiness limitations

0.4.1 Repairs

Repair or replace damaged parts prior to next flight. Follow the instructions of the DG-1000 repair manual for repairs of the airframe. Repairs outside the scope of DG-1000 repair manual and major repairs must be accomplished at a certified repair station or by a certified mechanic rated for composite aircraft structure work in accordance with DG repair methods.

Use only genuine spare parts.

For all aircraft under EASA regulations the following applies: According to part 21, subpart M to accomplish major repairs an approved repair instruction is required, see also TN DG-G-01 "Approved repair methods according to EU Commission Regulation 1702/2003 part 21, subpart M"

0.4.2 Life time of the airframe

The maximum allowable operating time for German composite sailplanes and motor gliders was proofed for 12000 flight hours. The initial life time for the DG-1000T is 3000 flight hours and may be increased by inspections according to section 2.4 of this manual to 6000 h, 9000 h, 10000 h, 11000 h and 12000 h.

0.4.3 Life time of components

Use only genuine spare parts. For part. No.'s of all parts please refer to section 8.

- a) The **gasket for the drainer valve** has to be exchanged after 6 years, part no. 60504402.
- b) The **spark plugs** have to be exchanged after 25 engine hours, part no.40050360.
- c) The **fabric straps of the safety harness** have to be exchanged according to the instructions of the respective manufacturer. If no limitations are given, exchange after 12 years, approved types see section 6.
- d) The **rubber cord** in the elevator control system see section 1.2.6 has to be replaced at least every 6 years, part no. 30091131.
- e) The adapter ring with elastic damping element 10M067 for the propeller (introduced with TN1000/26) has to be exchanged after 50 engine hours or 5 years, whichever comes first.
- f) The eccentric axle of the reduction gear (propeller axle) Solo part no. 2031211-V2 has to be replaced after 50 engine hours

Note: The **brake fluid of the wheel brake** has to be exchanged after 4 years (types see section 1.6.2).

Note:All **other components** like propeller, tow hook, wheels, gas struts, control system parts, bolts, pins etc. have no life time limitation, but should be replaced when worn, damaged or disqualified by excessive corrosion.

0.4.4 Service time, maintenance documents

Follow the instructions of the respective manufacturer:

- a) Operating Manual for Safety Tow Releases Series: Europa G 88 Safety Tow Release latest approved version
 Operating Manual for Tow Releases Series: E 85 Nose Tow Release latest approved version
- b) Safety harness: instructions of the manufacturer latest approved version. Approved types see section 6.
- c) Minimum instrumentation: instructions of the manufacturer, approved instruments see sections 6.
- d) Engine: Manual for the engine SOLO Type 2350 C, latest approved version.
- e) Operating- and Service Instructions for propeller DG-P001 latest approved version.

Note only for USA: The Airworthiness Limitations section is FAA approved and specifies maintenance required under Secs. 43.16 and 91.403 of the Federal Aviation Regulation unless an alternative program has been FAA approved.

I

2 Inspections

2.1 Daily inspection

see flight manual section 4.3

2.2 Regular inspections

A Annual inspection (and 100hr inspection – only for USA)

- Execute all items of the daily inspection see flight manual section 4.3.
- Check the rudder cables for wear especially around the "S" tubes on the rudder pedals. Worn rudder cables should be replaced (see section 4.2).
- Check the seals of the rudder (see section 1.3.5).
- Inspect all bolted connections and locking devices ie. locknuts, split pins etc.
- Check all metal parts for adequate greasing and rust prevention. (see section 3.3).
- Check the control surface deflections (see sections 1.2 up to 1.4).
- Check the free play in all control circuits (see section 1.2 up to 1.6)
- Check the fore and aft play of the wings (see section 1.11).
- Check the canopy emergency releases according to section 7.14 of the flight manual.
- Check the tension of the lines of the waterbag attachment (see section 4.1).
- Check the rubber cords in the control system (see sections 1.2.6 and 1.7.5.
- Check the thickness of the wheel brake linings and of the brake disc (see section 1.6.4).
- Check if the brake fluid has to be exchanged (see section 1.6.4).
- Check the airbrakes according to section 4.4.
- Check the fin ballast tank system according to section 1.8.2.
- Check the fin ballast box according to section 1.9.
- Check the friction of the canopy opening handles (canopies removed from fuselage): A force of 10 20N (2.2 up to 4.4 lbs.) should be required at the end of the handle. If the force is too low tighten the hinge bolt of the handle accordingly.
- Check if the powerplant has been serviced according to section 3.6.1.
- Check the friction brake of the throttle control (see section 1.12.8).
 Option: throttle handle in rear cockpit TN1000/15: Check of friction brake not applicable, not installed.
- Check the torque of the propeller bolts (see section 3.6.1 item 23).
- **Tow hooks:** The operating and maintenance instructions for the release mechanisms, see sect. 0.4.4 of this maintenance manual have to be followed.
- All-up weight and centre of gravity: These should be checked at least every 4 years during the yearly inspection.

A Special inspections

C.G. Tow hook:

After a wheel-up landing, the C.G. tow hook is to be cleaned and to be carefully checked for any damage.

Fuselage nose:

After a landing where the fuselage nose has touched the ground, the nose tow hook is to be cleaned and to be checked for correct functioning.

Clean the hole of the PC port (necessary for the stall warning) located behind the fuselage nose on the lower surface.

C.G. weighing: After all work which may influence the C.G.

8. Check the intake air filters of the carburettors for excessive dirt and wear, wash with pure petroleum spirit and blow compressed air in reverse direction through the filter. Spray the outside with oil for filters with cotton fabric, reinstall the filter.

We recommend exchange of the filters every 25 hours. Also new filters must be sprayed with filter oil.

- 8.a With the airfilters still removed check visually the screws of the throttle and choke valves for tight fit.
- 9. Clean the engine.
- 10. Check the muffler for cracks and ensure mounting is secure.
- 11. Check all engine nuts and bolts with a torque wrench (see sect. 1.12.10).
- 12. Check the rubber engine mounts, especially for cracks. Therefore apply strong pressure to the engine in forward, backward and sideways direction.
- 13. Check and grease the starter motor gear shaft (don't grease the starter motor gear) Check starter motor for tight mounting. There should be no excessive radial free play of the starter motor gear axle. With too much free play the starter must be exchanged.
- 14. Clean the starter ring gear and check for damage. Check if the starter ring gear was bent forwards by the starter motor.
- 15. Check the drive belts for wear and correct tension (see sect. 4.10.1). If a drive belts shows signs of wear the drive belt must be replaced.
- 16. Clean the spindle drive, function check. Check the connection of the spindle drive to the powerplant.
- 17. Check the propeller stopper for wear and function, check especially the rubber buffer for wear.
- 18. Check all the hinges on the engine compartment doors for proper fit and any cracks, tears etc. Check if hinge pins and the engine door spring are secured properly.
- 19. Oil all hinge points of the powerplant
- 20. Check the time taken to extend the power plant. If it takes longer than described under sect.1.13.2 the gas strut has to be replaced.

Variometer					
Manufacturer	Туре	Certification No.			
Winter	5 StVM5 (diam. 58)	TS 10.230/14			
	\pm 5 m/s Ident.No. 5451				
	<u>+1000 ft/min Ident.No. 5452</u>				
	\pm 10 kts Ident.No. 5453				
Winter	5 STV 5 (diam. 80)	TS 10.230/13			
	\pm 5 m/s Ident.No. 5251				
	<u>+</u> 1000 ft/min Ident.No. 5252				
	\pm 10 kts Ident.No. 5253				
Turn and bank indicator					
Manufacturer	Туре	Certification No.			
Apparatebau					
Gauting	WZ-402/31 12 V	10.241/8			
Accelerometer					
(for Category A A	erobatics)				
Accelerometer capable of retaining max. and min. g-values with markings					
red radial lines at $+7g$ and $-5g$.					
Manufacturer	Туре	Certification No.			
Falcon Gauge	GM5 10-2	MIL-A-5885 C			
AOA	BM 470-RL/L	MIL-A-5885 A			

Outside air temperature gauge

incorporated in the DEI-NT see below

Engine instrumentation

(RPM, fuel, CHT, voltmeter, engine elapsed time, outside air temperatureManufacturer TypeDG Flugzeugbau DEI-NT-DG-1000T

8.2 Parts for fuel system

- 60507550 Drainer CAV 110 (1/8" NPT)
- **Warning:** Replace the seal ring of the drainer against part no. 60504402 prior to installation
- 60504402 Seal ring for drainer CAV 110 (for automotive fuel)
- 60507560 Electric fuel pump Facet 40105
- 60507558 Refuelling pump KAVAN 12 V up to ser. no. T28
- 60507562 Refuelling pump Facet 60106 from ser. no. T29 on
- 60500164 Mechanical fuel pump Bing 8080 (no more available)
- 60500257 Mechanical fuel pump Mikuni DF44-18 from ser. no. T57 on and as spare part (for installation follow TN1000/28)
- 60507571 MANN-fuel- filter 500009180 WK 31/2(10) for refuelling pump
- 41070521 Fuel distributor 10M52/1 with filter for primer valve and restriction for excess fuel line
- 60000527 Fuel cock KH 1072 T
- 60507609 Coupling for fuel filler hose KL-006-0 WR513
- 60503070 Primer-valve IWP069
- 45001576 Full tank sensor with wiring, plug and gasket 60507547
- 60507547 Gasket O-ring 10 x 2,5 80FPM610 for full tank sensor
- 60000103 Fuel hose PU hydrolyse and microbe-resistant 6x1,5x9 mm
- 60000102 Fuel hose PU hydrolyse and microbe-resistant 8x2x12 mm

8.3 Parts for the electrical system

Battery 12V 17Ah 60510898 41076000 DEI-NT-DG1000T DEI-NT-DG1000T second unit (rear cockpit) 41076020 41076010 Control unit-NT 10E601 60510815 Limit switch 164-025 05 for positions engine extended and for stopping the engine Limit switch for propeller-stopper 164-564 60510463 Limit-switch engine retracted and engine extended 164-574, 60510464 alternatively SI2010-B2T20YR30,5m Manual extension-retraction switch MTG 206 S 60510484 Key switch 3 Pos, 2 Pol KL09-1908KA with wiring (Master 60510859 switch) Switch STA 106 E (selector intern-extern) 60510362 60510372 Press-button DJET 07.17502.21 for starter 60510375 Press-button 12G2904 for refuelling pump Circuit breaker ETA 2A 60510385 Circuit breaker ETA 3A 60510386 60510437 Fuse 80 A for main battery Fuse socket for main battery fuse 80 A 60510434 Proximity switch 60510550 Proximity switch 10E107 ready assembled with wiring and plug 41075000 10002317 Preh plug for 12V sockets From ser. No. 10-120 on: 60510796 Socket BSB12 (in main bulkhead) Plug BSK12 (for socket BSB12) 60510797

From ser. No. 10-202 on:

- 60510880 Socket XLR 3-pole NC3FD-LX-BAG (in main bulkhead)
- 60510881 Plug XLR 3-pole NC3MX-BAG (for socket XLR)

