Rev.	Affected	Description	Issue	EASA	Inserted
No.	Pages/		Date	Approval	Date
	section			Date	Signature
7	0.3 0.6, 0.7,	Electrically operated	Nov.	28 January	
	9.1-9.12	main landing gear	2008	2009	

0.2 List of effective pages

Section		page	issued	replaced	replaced	replaced	replaced
0		0.0 0.1 0.2 0.3 0.4 0.5 0.6 0.7	July 2005 see m	nanual amen " " " "	dments		
1		1.1 1.2 1.3 1.4 1.5 1.6	""	Jan. 07 Jan. 07			
2	App. "" "" "" "" "" "" ""	2.1 2.2 2.3 2.4 2.5 2.6 2.7 2.8 2.9 2.10 2.11 2.12 2.13 2.14 2.15	July 2005	Jan. 07 Jan. 07 Jan. 07 Jan. 07 Jan. 07	May 2008 Oct. 07		
3	"" "" "" "" "" "" "" "" "" "" "" "" ""	3.1 3.2 3.3 3.4 3.5 3.6 3.7 3.8	July 2005	Jan. 06			

0.2 List of effective pages (cont.)

Section	Page	issued	replaced	replaced	replaced	replaced
8	8.1	July 2005				
	8.2	"				
	8.3	"				
	8.4	"				
	8.5	"				
	8.6	"				
9	9.1	Nov. 2008				
	9.2	"				
	9.3	"				
	9.4	"				
	9.5	"				
	9.6	"				
	9.7	"				
	9.8	"				
	9.9	"				
	9.10	"				
	9.11	**				
	9.12	"				

0.3 Table of contents

	Section
General (a non-approved section)	1
Limitations (an approved section)	2
Emergency procedures (an approved section)	3
Normal procedures (an approved section)	4
Performance (a partly approved section)	5
Mass (weight) and balance (a non-approved section)	6
Motorglider and systems description (a non-approved section)	7
Motorglider handling, care and maintenance (a non-approved section)	8
Supplements	9

Issued: July 2005 0.7

9 Supplements

Section		page
9.1	Introduction	9.2
9.2	List of inserted supplements	9.2
9.3	Electrically operated main landing gear	9.3 – 9.11

9.1 Introduction

This section contains the appropriate supplements necessary to safely and efficiently operate the sailplane when equipped with various optional systems and equipment not provided with the standard sailplane.

9.2 List of inserted supplements

Date of	Document No.	Title of the inserted supplement
insertion		
November	Section 9.3	Electrically operated main landing gear
2008		TN1000/14

9.3 Electrically operated main landing gear

Introduction

In the following text the changes to those sections of the flight manual which are affected by the installation of the electrically operated main landing gear will be given

Section 1 General

1.4 Descriptive data

Subsection amended

With landing gear version A) the main landing gear may be operated electrically as an option, Version D).

D) Very high, spring mounted, electrically operated retractable main landing gear, wheel with hydraulic disc brake, tail wheel

With this version there are no handles and control rods for manual operation of the landing gear like Version A.

A landing gear warning device is integrated into the system.

Both cockpits are equipped with all controls and control lights

A manually operated emergency extension system is provided.

Technical data:

Extension and retraction time (electrically operated): approx. 7seconds

Extension time emergency operation: approx. 2 seconds

Section 2 Limitations

subsections amended

2.2 Airspeed

Note: VLO is not changed compared to the manually operated main landing

gears

	Speed	IAS	Remarks
		km/h	
		(kts.)	
VLO	Maximum landing	185	Do not extend or retract the landing gear
	gear operating	(100)	above this speed.
	speed		
VLE	Never exceed speed	150	Do not exceed this speed in case the landing
	in case the landing	(81)	gear is not locked, see emergency
	gear is not locked		procedures sect. 3.20

Warning: If the landing gear is operated at speeds higher than VLO and if gusts generate accelerations higher than 4 g the landing gear may be damaged.

2.9 Approved manoeuvres

No aerobatic manoeuvres are allowed during extension and retraction of the landing gear and in case you fly with the landing gear is unlocked (see emergency procedures sect. 3.20).

2.10 Manoeuvring load factors

	max. speed	g-load
operating the landing gear	VLO	+4
		-2,65
in case the landing gear is not locked	VLE	+4
		-2,65

2.21 Limitation placards

Below the controls and control lights for the electrically operated landing gear see sect. 7.3:

LG ext.-retr. up to 185 km/h 100 kts.

Section 3 Emergency procedures

new subsection

3.19 Emergency extension of the electrically operated main landing gear

If extending the landing gear via the electrical system is not possible, the landing gear may be extended manually. The extension force is produced by a gas strut. For emergency extension pull on one of the 2 black-red emergency extension handles (located at the left hand fuselage wall, one in each cockpit), pull the handle until the landing gear is fully extended. The travel is about 15 mm (.6 in.), the extension time about 2 seconds. During extension the centre (red) LED is shining. When the landing gear is fully extended the lower green LED also starts shining.

Warning: Don't let the handle go before the green LED starts shining. If you don't pull for long enough the landing gear will rest in a position not locked over centre and will be destroyed at touch down.

Caution: If you execute the emergency extension at high airspeed it may take a longer time until the landing gear is fully extended. You should shorten the time by reducing the airspeed.

Caution: If emergency extension was necessary check the system to detect the failure and repair the system.

Resetting the system for normal operation see section 4.5.12.3.

3.20 Incompletely retracted electrically operated main landing gear

If the landing gear is not locked the centre (red LED) starts blinking approx. 22 seconds after the start of the retraction.

In this condition the retraction mechanism may be damaged with g-loads exceeding 4 g.

Instruction 1: Extend and retract the landing gear again.

Instruction 2: If instruction 1 was not successful extend the landing gear and land as soon as possible to fix the problem. For the remainder of the flight, don't fly faster than 150 km/h (81 kts.) and avoid abrupt manoeuvres.

Section 4 Normal procedures

4.2 Rigging and derigging

new subsection

4.2.8 Extension and retraction of the electrically operated main landing gear

Extension see 4.5.12.1 normal procedures

Warning: The main landing gear can only take up the weight of the glider when fully extended and locked. Therefore it is esential that during extension of the landing gear no load is applied to the main wheel. Lift the trailer ramp high enough. If there is a risk that the ramp may come down while the landing gear is extending secure the ramp, e.g. with blocks.

If such a case happens nevertheless switch the landing gear to retraction. To accomplish this you must switch up and hold the toggle switch and press the press button simultaneously 3 times within 2 seconds.

Warning: If you operate the retraction or extension the landing gear will travel to the up or down stop. So make sure that there are no obstacles which may be caught by the landing gear to prevent damage or injuries.

You can stop the travelling by switching the toggle switch in the reverse direction and pressing the press button simultaneously

4.5 Normal procedures

new subsection

4.5.12 Electrically operated main landing gear

4.5.12.1 Extension and retraction in flight

Retraction: For retraction switch and hold the toggle switch up and press the press button twice within 2 seconds. With each press on the button a signal will sound see section 4.5.16. The landing gear will retract automatically. You may let go of the switches. During retraction the centre (red) LED will shine and the upper green LED will blink. As soon as the landing gear is retracted and locked only the upper green LED will shine.

Warning: If the upper green LED doesn't start to shine and the red LED instead starts blinking refer to section 3.20 emergency procedures.

Extension: For extension switch the toggle switch down and let go.. The landing gear will be extended and locked.

During extension the centre (red) LED will shine and the lower green LED will blink. As soon as the landing gear is extended and locked only the lower green LED will shine.

Note: In case of high acceleration during extension or retraction an over current cut off system will switch off the spindle drive to protect the system. As soon as the g-loads decreasa, the landing gear will continue to travel.

Note: To save electrical power during flight the upper green LED will stop shining after approx. 5 minutes, landing gear retracted and locked.

4.5.11.2 Extending the landing gear via the emergency extension system.

The emergency extension system is also designed to be operated for in flight training purposes. Operation see section 3.19.

Resetting the system for normal operation should be executed after landing, for procedure see section 4.5.12.3.

Caution: It is strongly recommended to train the emergency extension in flight.

Note: Resetting the system for normal operation is also possible in flight.

However, this is only permissible if there are 2 pilots on board, one pilot flying the glider and the other resetting the system.

Then you may retract the landing gear again according to section 4.5.12.1. to continue the flight.

4.5.12.3 Resetting the emergency extension system for normal operation

After an emergency extension the system must be reset for normal operation. To accomplish this you must pull one of the 2 emergency extension handles and simultaneously switch the toggle switch down. The centre (red) and the lower green LED will shine.

Switch and handle must be operated until the centre LED stops shining and only the lower green LED continues shining. The spindle drive will then stop operating, then let go handle and switch

Thereafter you may retract the landing gear again according to section 4.5.12.1.

4.5.12.4 Part extension and retraction for inspection and servicing

The retraction may be stopped by switching the toggle switch down, The extension may be stopped by switching the toggle switch up and pressing simultaneously the press button.

Only the red LED will shine.

For any service work switch off the main switch!

With the procedures described in section 4.5.12.1 you may retract or extend the landing gear again.

4.5.12.5 Precautionary measures against retracting the landing gear while on the ground

If the glider is resting on the main landing gear the landing gear should not be retracted, as retraction will result in damage of the landing gear. To minimise the risk of such operating error the following safety features have been incorporated:

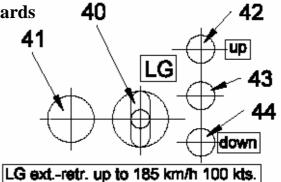
- 1. If the toggle switch is switched up, nothing will happen.
- 2. If the toggle switch is switched up and the press button is pressed 1 time a warning tone will sound, otherwise, nothing will happen.
- 3. Only if the toggle switch is switched up and the press button is pressed 2 times within 2 seconds the landing gear will be retracted.

Caution: If you leave the DG-1000 unattended switch off the main switch to prevent any operating error.

Section 7 Sailplane and system description

7.3 Cockpit, cockpit controls and placards

Subsection amended



40) Toggle switch for extension and retraction of the electrically operated main landing gear

up retracted and locked

down extended and locked over centre

41) Press button for retraction of the electrically operated main landing gear For retraction switch and hold the toggle switch up and press the press button 2 times within 2 seconds.

42) - 44) control lights (LED's) for the electrically operated main landing gear:

control light indication
42) upper green shines
44) lower green shines
43) red shines, 42) upper green blinking
43) red shines, 44) lower green blinking
43) red shines
43) red shines
44) lower green shines, 43) red shines
44) lower green shines, 43) red shines
43) red blinking + upper or lower green
blinking according to travel direction
all LED's blinking

Note: To save electrical power during flight the upper green LED will stop shining after approx. 5 minutes if the landing gear is retracted and locked.

45) Landing gear emergency extension handles black-red The handles are located at the left hand fuselage wall, one in each cockpit at the positions of the handles for the manually operated landing gears, item 10).



For emergency extension pull on one of the 2 red emergency extension handles, pull the handle until the landing gear is fully extended.

Note: 26) Control light for the trim ballast box in the fin may not be installed at the position shown in the sketch on page 7.3 but at another suitable place in the front instrument panel.

7.7 Landing gear

Subsection amended

D) Very high, spring mounted, electrically operated retractable main landing gear, wheel with hydraulic disc brake, see diagrams 20 and 21 M.M, tail wheel With this version there are no handles and control rods for manual operation of the landing gear like Version A.

In the normal operating mode the landing gear will be retracted and extended by an electrical spindle drive.

A control unit which is installed in the rear instrument tower controls all electrical functions and the control lights.

A landing gear warning device is integrated into the system.

Both cockpits are equipped with all controls and control lights

The landing gear will be locked in the extended position by over centre locking of the drag struts.

The landing gear will be locked in the retracted position by 2 latches.

The system is equipped with a safety circuit against retraction of the landing gear on the ground see section 4.5.12.5.

The system is equipped with an over current cut off which stops the extension or retraction if high accelerations occur to protect the drive against damage. As soon as the g-loads lower, the landing gear will continue to travel.

Emergency operation: If the electrical system is damaged or no battery power is available, the landing gear may be extended manually. The handles are located at the left hand fuselage wall, one in each cockpit at the positions of the handles for the manually operated landing gears).

Pulling on one of the 2 red emergency extension handles will open the valve of a lockable gas strut. The gas strut will push the spindle drive forward on a linear guide to extend and lock the landing gear.

Technical data:

Extension and retraction time (electrically operated): approx. 7seconds

Extension time emergency operation: approx. 2s

Power: 12V, max. 10 A

7.14 Electrical system

Subsection amended

Electrical system with electrically operated landing gear

Wiring see wiring plan 10E4 enclosed to the maintenance manual.

Fuses:

The electrically operated landing gear is protected by an automatically reset fuse in the landing gear control unit.

Landing gear warning:

A landing gear warning device is integrated into the system. Warning is by a buzzer and in addition via the . DEI-NT see section 7.4.5.