# 0 Manual Contents

# 0.1 Log of Revisions

Rev	Pages	Reference	Rev.
no.			date
1	0-1, 0-3, 0-4, 0-5, 0-6, 0-8, 0-11,	TN 8017, necessary changes to	Nov.
	1-27, 3-4, 4-16, 4-19, 4-21,	the power plant	2010
	10-2, 10-8, 11-2, 11-18, 11-20		
2	0-1, 0-3, 0-7, 1-1, 1-7,	TN8019, wheel brake actuated by	Feb.
	1-11, 1-12, 8-3	airbrake handle.	2011

## **0.2 List of Effective Pages**

Chapter	Page	Edition	Replaced	Replaced	Replaced
0	Title page	April 2005			
	0-1	April 2005	Feb. 2011		
	0-2	April 2005			
	0-3	April 2005	Feb. 2011		
	0-4	April 2005	Nov. 2010		
	0-5	April 2005	Nov. 2010		
	0-6	April 2005	Nov. 2010		
	0-7	April 2005	Feb. 2011		
	0-8	April 2005	Nov. 2010		
	0-9	April 2005			
	0-10	April 2005			
	0-11	April 2005	Nov. 2010		
	0-12	April 2005			
	0-13	April 2005			
1	1-1	April 2005	Feb. 2011		
	1-2	April 2005			
	1-3	April 2005			
	1-4	April 2005			
	1-5	April 2005			
	1-6	April 2005			
	1-7	April 2005	Feb. 2011		
	1-8	April 2005			
	1-9	April 2005			
	1-10	April 2005			
	1-11	April 2005	Feb. 2011		
	1-12	April 2005	Feb. 2011		
	1-13	April 2005			
	1-14	April 2005			
	1-15	April 2005			
	1-16	April 2005			
	1-17	April 2005			
	1-18	April 2005			
	1-19	April 2005			
	1-20	April 2005			

## **<u>0.2 List of Effective Pages</u>** (continued)

Chapter	Page	Edition	Replaced	Replaced	Replaced
5	5-1	April 2005			
	5-2	April 2005			
	5-3	April 2005			
	5-4	April 2005			
	5-5	April 2005			
	5-6	April 2005			
	5-7	April 2005			
	5-8	April 2005			
	5-9	April 2005			
	5-10	April 2005			
	5-11	April 2005			
	5-12	April 2005			
	5-13	April 2005			
	5-14	April 2005			
	5-15	April 2005			
	5-16	April 2005			
	5-17	April 2005			
	5-18	April 2005			
6	6-1	April 2005			
	6-2	April 2005			
	6-3	April 2005			
	6-4	April 2005			
7	7-1	April 2005			
0	0.1	4 :1.200.5			
8	8-1	April 2005			
	8-2	April 2005	F 1 2011		
	8-3	April 2005	Feb. 2011		
9	0.1	April 2005			
9	9-1	April 2005			
	9-2	April 2005			
	9-3	April 2005			

### 1. SYSTEM DESCRIPTION AND ADJUSTMENT DATA

### 1.1 Summary

## Wings

Wing span variable by exchange of 15m winglets against 18m tips with winglets.

### **Aileron Controls**

Aileron system activated via pushrods, connection of system by automatic coupling during rigging. Dynamic aileron mass balance in wing, aileron at 18m wingspan in two parts.

## **Elevator** Controls

Elevator system activated via pushrods, automatic coupling of system during rigging. 100% mass balance in vertical tail fin pushrod.

### **Rudder Controls**

Rudder system activated via cables, 100% mass balance at rudder.

### Wheel Brake

Feet operated, activated by Bowden cable from rudder pedals.

**With TN8019 executed**: The wheel brake Bowden cable is connected to the airbrake drive instead of the rudder pedals.

### Air Brakes

Activated via pushrods. Automatic connection of system during rigging. Locking mechanism in wings. Upper surface double height air brakes with spring loaded cover blades. Friction damper in box to prevent oscillations during extension.

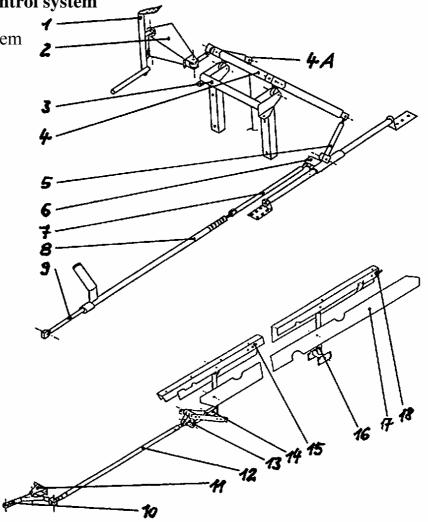
## Water Ballast System

Two integral tanks per wing, maximum capacity per wing 95 Litres <25.1 US gallons, 20.9 Imp. gal.>. Two valves at under side of wing for loading and dumping operated by one root rib pin. Automatic connection during rigging. Ventilation of both wing integral tanks at root-rib.

Ballast tank in the vertical tail fin for mandatory C.G. movement compensation due to wing water ballast as well as compensation for weight of heavy pilots, capacity 7.5 Litres <1.98 US gallons, 1.65 Imp.gallons>. Maximum permissible compensation allowed for in tables.

## 1. System Description and Adjustment Data (continued)

# 1.3 Air Brake Control system 1.3.1 Control System



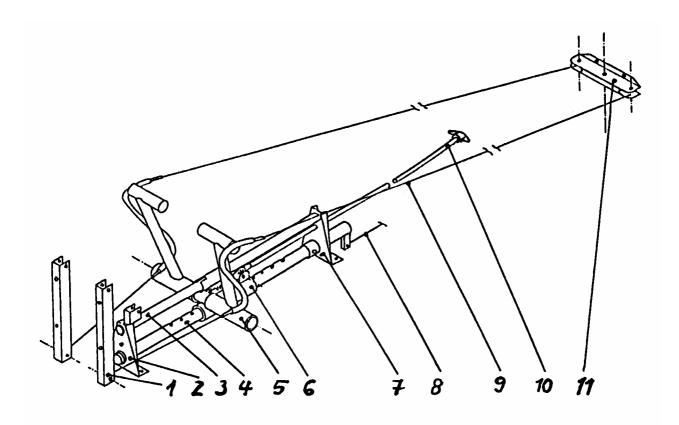
<u>Fuselage</u>			Wings		
No.	Denomination	Drawing	No.	Denomination	Drawing
1	Automatic connector	3R10-119	10	Root rib drive lever	3F4-64
	bearing			or	3F4-71
2	Automatic air brake	3R6-44	11	Root rib bracket	4F3-76
	connector at fuselage		12	Wing pushrod	4F4-63
3	Air brake drive bracket	3R6-78	13	Locking lever	4F4-53
4	Air brake drive	3R6-50	14	Inner lever	3F4-51
With TN8019 executed:		15	Upper blade	3F4-60	
4	Air brake drive shaft	9St13	16	Outer lever	3F4-52
4A	Wheel brake cable		17	Lower blade	3F4-54
	attachment				
5	Intermediate rod	4R10-77	18	Friction brake	
6	Sliding connector	4R6-47			
7	Air brake rod	4R6-59			
8	Air brake handle	3R6-74			
9	Handle guide tube	4R6-31			

## 1. System Description and Adjustment Data (continued)

# 1.5 Rudder Control System, Wheel Brake

# 1.5.1 Control System

No.	Denomination	Drawing	
1	Canopy opener bracket	4R8-67	
2	Forward pedal guide bracket	3R14-14	
3	Upper pedal guide tube	4R14-18	
4	Lower pedal guide tube	4R14-19	
5	Rudder pedal	1R14-21	
6	Pedal support	3R14-16	
7	Rear pedal guide bracket	3R14-15	
8	Wheel brake cable		
8	Not installed with TN8019 executed		
9	Rudder cable		
10	Pedal adjustment cable	4R14-31	
11	Rudder drive bracket	4S1-10	



## 1. System Description and Adjustment Data (continued)

### **1.5 Rudder Control System, Wheel Brake** (continued)

### 1.5.2 Deflections and Tolerances

<u>Rudder</u> : 10 both sides 20° - 30°	Rudder:	To both sides	26° - 30°
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For easier checking, measured angles may be converted to mm / in deflection values, using the actual local radius of the defined measuring place. See also table below.

### Limit values for aileron deflections in Millimetres/Inches

Rudd	Rudder				
Mea	Measured value = Distance from lower				
	trailing edge to centre of cable bolt				
Measured		Local	26° to 30°		
Value		Radius			
mm / in		mm / in	mm / in		
395		390	175 to 202		
	15.551	15.354	6.890 to 7.953		
396		391	176 to 202		
	15.591	15.394	6.929 to 7.953		
397		392	176 to 203		
	15.630	15.433	6.929 to 7.992		

## 1.5.3 Stops

Rudder stops are fibre glass plates at lower rudder end against vertical tail shell.

### 1.5.4 Wheel Brake

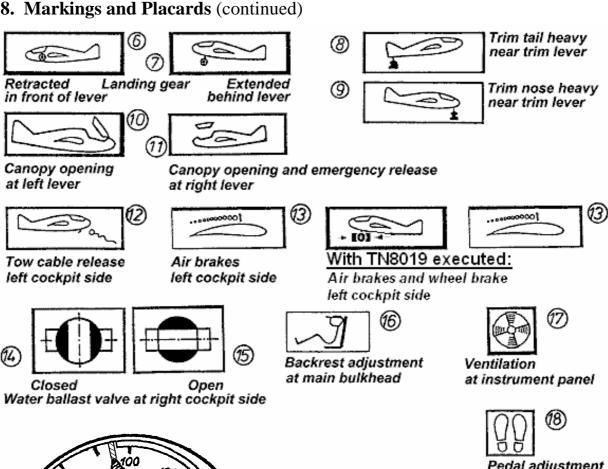
Wheel brake adjustment by Bowden cable adjusting screws at both cable ends –under seat and at landing gear wheel fork.

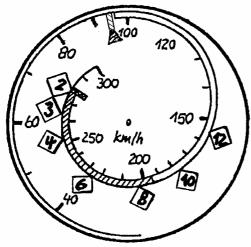
Stop for wheel brake operation at lower guide tube of pedal support.

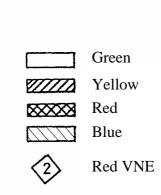
### With TN8019 executed:

Wheel brake adjustment by Bowden cable adjusting screws at both cable ends – at the back of the landing gear box and at landing gear wheel fork.

### **8. Markings and Placards** (continued)







Pedal adjustment at lower edge of instrument panel